

**Electromagnetic Wave Propagation for Space-borne Systems – Prof. L. Luini,  
September 5<sup>th</sup>, 2025**

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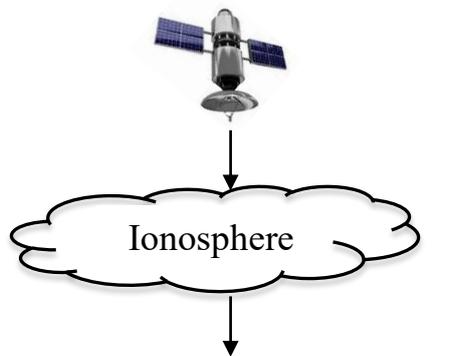
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**Problem 1**

An Earth observation satellite, at altitude  $H = 800$  km, downloads data to a ground station using two channels centered around the carrier frequencies  $f_1 = 25$  MHz and  $f_2 = 35$  MHz (zenithal link).

- 1) Determine  $T_2$ , the data travel time at frequency  $f_2$ , knowing  $T_1 = 2.774$  ms (data travel time at frequency  $f_1$ ).
- 2) If the ionospheric profile changes due to a sudden ionospheric anomaly, determine the maximum peak electron content value,  $N_{max}$ , for the system to operate correctly.

Assumption: no effects induced by the troposphere (neither delay, nor attenuation).



**Solution**

1) The total travel time  $T$  is due to the free space and to the ionosphere, the latter being frequency dependent.  $T$  is defined as:

$$T = T_{FS} + T_{IONO} = \frac{H}{c} + \frac{1}{2c} \frac{81}{f^2} TEC$$

where  $TEC$  is the total electron content. As a result, inverting that equation with  $T_1$ ,  $f_1$  and  $H \rightarrow$   
 $TEC = 49.7$  TECU

Therefore, using the same equation,  $T_2 = 2.721$  ms.

2) For the system to operate correctly, both channels need to download data to the ground station. When  $N_{max}$  increases, so does the critical frequency  $f_c$ . Specifically, both carrier frequencies need to be higher than the critical frequency for the waves to cross the ionosphere (zenithal pointing). Therefore:

$$f_c = 9\sqrt{N_{max}} < f_1 \rightarrow N_{max} < \frac{f_1^2}{81} = 7.72 \times 10^{12} \text{ e/m}^3.$$

## Problem 2

A pulsed radar with zenithal pointing (transmit power  $P_T = 1$  kW), working with carrier frequency  $f = 80$  GHz and with antenna gain  $G = 40$  dB, illuminates an aircraft flying at high altitude. As depicted in the figure below, the wave crosses a layer consisting of anisotropic particles ( $h_T = 3.5$  km), whose propagation constants are:

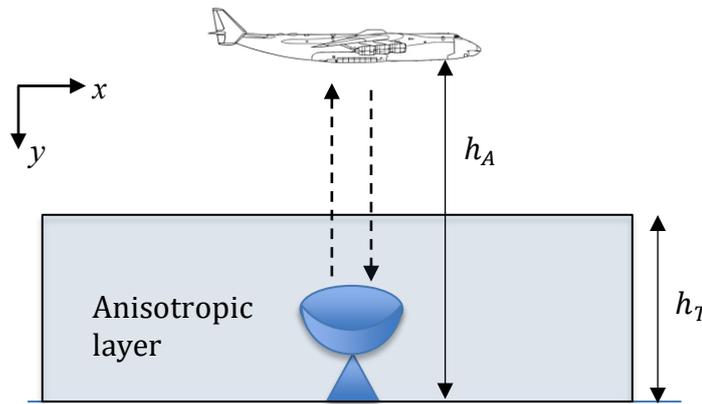
$$\gamma_z = \alpha_z + j\beta_z = 6 \times 10^{-4} + j1675.5 \text{ 1/m}$$

$$\gamma_x = \alpha_x + j\beta_x = 4 \times 10^{-4} + j1675.5 \text{ 1/m}$$

For this scenario:

- 1) Which polarization among the following ones allows improving the radar sensitivity to small aircrafts? Linear along  $x$ , linear along  $z$ , circular.
- 2) Working with the polarization chosen at point 1, calculate the minimum backscatter value  $\sigma$  that allows the target identification by the radar, knowing that the radar receiver sensitivity is  $P_m = 1.279 \times 10^{-15}$  W. Consider all targets flying at altitude  $h_A = 8$  km.

Assumption: no further tropospheric attenuation other than that due to the anisotropic particles.



## Solution

1) From the propagation constants, it emerges that the two linear polarizations are subject to the same phase delay, but to a different attenuation: the  $z$ -component will be more attenuated than the  $x$ -component. If the radar pulse is less attenuated, it can identify smaller targets (i.e. associated to a lower backscatter  $\sigma$ )  $\rightarrow$  it is better to use the linear polarization along  $x$ .

2) Considering this scenario, the power density reaching the aircraft is:

$$S_A = \frac{P_T}{4\pi h_A^2} A_R G$$

The power received back by the radar is:

$$P_R = \frac{S_A \sigma}{4\pi h_A^2} A_R G \frac{\lambda^2}{4\pi}$$

The tropospheric attenuation  $A_R$  is given by:

$$A_R^{dB} \alpha_x \cdot 8.686 \cdot 1000 \cdot h_T = 12.16 \text{ dB} \rightarrow A_R = 0.0608$$

Inverting the equation by setting  $P_R = P_m$ , and solving for the minimum backscatter  $\rightarrow \sigma = 2 \text{ m}^2$ .

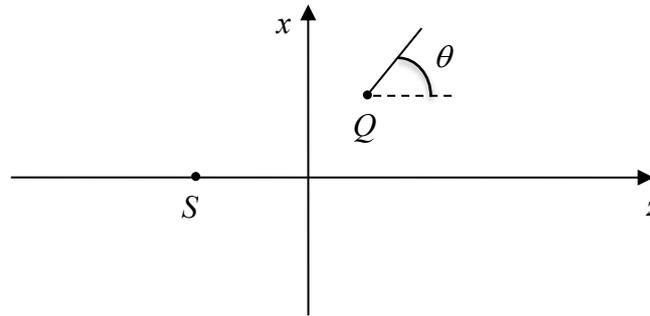
### Problem 3

A uniform plane wave (frequency  $f = 100$  MHz) propagates along the  $z$  axis in a perfect dielectric with  $\epsilon_{r1} = 9$  and  $\mu_{r1} = 4$ , and hits the surface of a medium characterized by conductivity  $\sigma = 5 \times 10^{-2}$  S/m,  $\epsilon_{r2} = 9$  and  $\mu_{r2} = 1$ . The incident electric field is polarized along  $x$  and its value in  $(0,0,0)$  is  $E_0 = 1$  V/m (see figure below).

Calculate:

- The electric field at point  $S(0,0,-2$  m).
- The power absorbed by the linear antenna located in  $Q(1$  m,  $1$  m,  $1$  m), lying on the  $xz$  plane ( $\theta = 45^\circ$ ).

Assume that both linear antennas have an effective area  $A_E = 2$  m<sup>2</sup>.



### Solution

a) For the second medium, the loss tangent is (no approximations possible):

$$\tan \delta = \frac{\sigma}{\omega \epsilon_{r2} \epsilon_0} = 1$$

The reflected and transmitted waves can be calculated through the reflection coefficient:

$$\eta_1 = \eta_0 \sqrt{\frac{\mu_{r1}}{\epsilon_{r1}}} = 251.3 \Omega$$

$$\eta_2 = \sqrt{\frac{j\omega\mu_2}{\sigma + j\omega\epsilon_2}} = 97.6 + j40.4 \Omega$$

$$\Gamma = \frac{\eta_2 - \eta_1}{\eta_2 + \eta_1} = -0.4214 + j0.1646$$

The electric field in  $S$  is the combination of the incident and the reflected waves:

$$\vec{E}(S) = E_0 \vec{\mu}_x e^{-j\beta_1 z_S} + \Gamma E_0 \vec{\mu}_x e^{j\beta_1 z_S} = 0.581 + j0.185 \text{ V/m}$$

$$\text{with } \beta_1 = \omega \sqrt{\epsilon_1 \mu_1} = 12.57 \text{ rad/m}$$

b) As the electric field has a linear polarization along  $x$  and the antenna has a  $45^\circ$  tilt, only part of the power density carried by the transmitted wave will be received. Specifically:

$$\gamma_2 = \sqrt{j\omega\mu_2(\sigma + j\omega\epsilon_2)} = \alpha_2 + j\beta_2 = 2.86 + j6.91 \text{ 1/m}$$

$$S(Q) = \frac{1}{2} \frac{|E_0(1 + \Gamma) \cos(\theta)|^2}{|\eta_2|} \cos^2(R \eta_2) e^{-2\alpha_2 z_0} \text{ W/m}^2$$

Therefore:

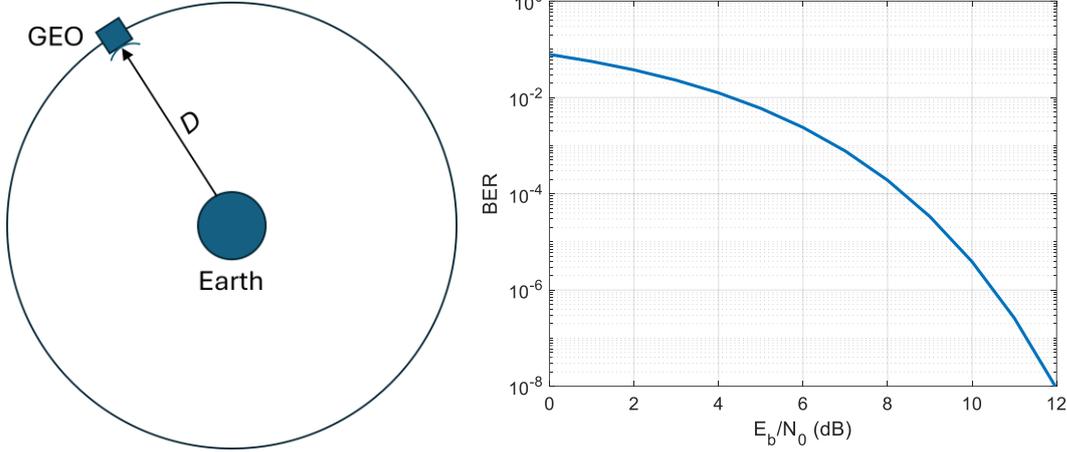
$$S(Q) = 2.6 \times 10^{-6} \text{ W/m}^2$$

Finally, the power received by the antenna is:

$$P(Q) = S(Q) A_E = 5.2 \times 10^{-6} \text{ W}$$

### Problem 4

Referring to the figure below, consider the uplink to a GEO satellite, that is seen at  $45^\circ$  elevation angle from the gateway on the Earth. The path length is  $D = 40000$  km and the two antennas are perfectly pointed. The gain of the satellite antenna and of the gateway antenna are  $G_S = 20$  dB and  $G_G = 34$  dB, respectively. The main beam of the satellite antenna covers an area with brightness temperature  $T_B = 200$  K.



1. Determine the BER of the link, considering the BER graph for the 2-PSK modulation reported above.
2. What is the maximum data rate that could be achieved for the conditions at point 1? How can this data rate be achieved?

Additional data:

- carrier frequency  $f = 18.1$  GHz
- power transmitted by the gateway:  $P_T = 1260$  W
- bandwidth of the receiver:  $B = 1$  MHz
- data rate:  $R = 1$  Mb/s
- internal noise temperature of the receiver:  $T_R = 100$  K
- CCDF of the zenithal tropospheric attenuation  $A$ :  $P(A) = 100e^{-1.15A}$  ( $A$  in dB and  $P$  in %)
- target service level:  $P_{AV} = 99.9\%$
- mean radiating temperature of the troposphere:  $T_{mr} = 260$  K.

### Solution

1) The signal-to-noise ratio (SNR) is given by:

$$SNR = \frac{P_R}{P_N} = \frac{P_T G_G f_G (\lambda/4\pi D)^2 f_S G_S A}{k T_{sys} B}$$

where  $k$  is the Boltzmann's constant ( $1.38 \times 10^{-23}$  J/K),  $T_{sys}$  is the total noise temperature (summation of  $T_R$  and the antenna noise temperature  $T_A$ ),  $f_G = f_S = 1$  (perfect pointing) and  $A$  is the tropospheric attenuation exceeded for  $P_{OUT} = 100 - P_{AV} = 0.01\%$  of the time. Inverting the CCDF expression and scaling the attenuation from zenith to the elevation angle  $\rightarrow A = 8.5$  dB = 0.1414. Regarding  $T_A$ , given the geometry, it is calculated as:

$$T_A = T_B A + T_{mr} (1 - A) = 251.5 \text{ K}$$

The resulting  $SNR$  is approximately 10 dB. Recalling that  $SNR = E_b R / N_0 B$ , we obtain  $E_b / N_0 = 10$  dB  
→  $BER \approx 4 \times 10^{-6}$ .

2) The maximum theoretical data rate is given by Shannon's formula:

$$C = B \log_2(1 + SNR) = 3.46 \text{ Mbit/s}$$

This data rate can be achieved by using complex information coding strategies (e.g. Turbo codes).