# EM wave propagation for space-borne systems – Prof. L. Luini, June 13<sup>th</sup>, 2024

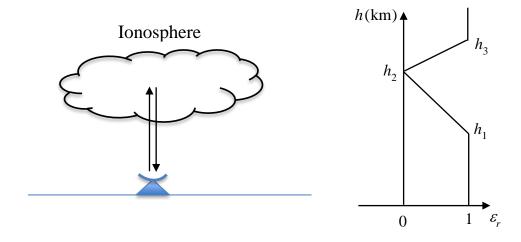
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## **Problem 1**

Making reference to the figure below, a monostatic ground-based pulsed radar system, operating at f=18 MHz and pointing zenithally, aims at monitoring the ionosphere, i.e. the peak electron content and its value. The trend of  $\varepsilon_r$  in the ionosphere is depicted in the figure below on the right side ( $h_1=100$  km). Knowing that the radar pulse round trip time is  $\tau=2.33169$  ms, determine the value and position of peak electron content  $N_{\text{max}}$ .

Assumption: neglect tropospheric effects.



# **Solution**

The peak electron content value can be obtained from the following expression:

$$cos(\theta) = \sqrt{1 - \left(\frac{9\sqrt{N_{\text{max}}}}{f}\right)^2} = \sqrt{\varepsilon_r}$$

Setting  $\theta = 90^{\circ}$  and inverting the expression  $\rightarrow N_{\text{max}} = 4 \times 10^{12} \text{ e/m}^3$ .  $N_{\text{max}}$  obviously corresponds to the lowest value of  $\varepsilon_r$ ; also, being  $\varepsilon_r = 0$  (for zenithal pointing) indicates that the wave is totally reflected exactly where  $N_{\text{max}}$  lies, i.e. at  $h_2$ .

The time required for the radar pulse to reach  $h_2$  from the ground is  $t = \tau/2 = 1.165845$  ms. Such time can be expressed as:

$$t = \frac{h_2}{c} + \frac{40.3}{cf^2} \text{TEC} = \frac{h_2}{c} + \frac{40.3}{cf^2} \frac{(h_2 - h_1) N_{\text{max}}}{2}$$

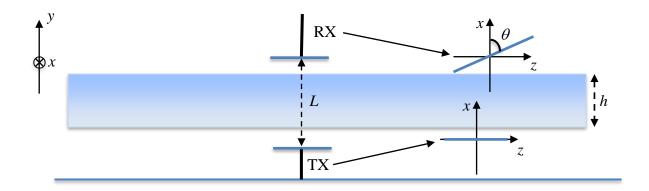
Solving for the only unknown, i.e.  $h_2 \rightarrow h_2 = 300$  km.

#### **Problem 2**

An Earth-space link, with path length L=400 km and operating at f=90 GHz, crosses a homogeneous cloud with thickness h=3 km. The transmitter (TX) uses a linear horizontal antenna, while the receiver (RX) linear horizontal antenna is tilted by an angle of  $\theta=60^{\circ}$  (see sketch below) due to problems with the satellite attitude control. For this link:

- 1) Determine the polarization of the wave in front of the receiver RX (specify: LH or RH for a circular/elliptical, tilt angle for a linear).
- 2) Calculate the power density in front of the RX antenna.
- 3) Calculate the power received by RX.

Assume: EIRP = 36 dBW; effective area of the RX antenna,  $A_{RX} = 2 \text{ m}^2$ ; specific attenuation due to clouds,  $\alpha_c = 2 \text{ dB/km}$ ; RX pointing at the Earth center and TX pointing at zenith; no additional losses at TX and RX.



#### **Solution**

- 1) As clouds consist of small spherical droplets, which are isotropic, the polarization transmitted by TX is unaffected by the presence of clouds.
- 2) The power density in front of the RX antenna is given by:

$$S_{RX} = \frac{EIRP}{4\pi L^2} f_T L_{TX} A_C \approx 0.5 \text{ nW/m}^2$$

where, given the assumptions:  $f_T = L_{TX} = 1$ .  $A_C$  is the cloud attenuation in linear scale. Such value in dB is obtained as:  $A = \alpha_C h = 6$  dB  $\rightarrow A_C = 0.2512$ .

3) The power received by RX needs to account for the RX antenna tilt:

$$P_{RX} = S_{RX}A_{RX}f_RL_{RX}[\cos(90 - \theta)]^2 \approx 0.75 \text{ nW}$$

## **Problem 3**

Consider the plane sinusoidal wave below, with f = 1 GHz and incident electric field given by:

$$\vec{E}_{i}(0,0,0) = j\mu_{x} \quad \text{(V/m)}$$

$$1 \quad \varepsilon_{r1} = 4 \quad y \quad \boxed{2} \quad \varepsilon_{r2} = 16 \quad \mu_{r1} = 4$$

$$\theta = 35^{\circ} \quad 0 \quad z$$

Calculate:

- 1) The total electric field in the first medium.
- 2) The power density in the second medium in the z direction.
- 3) What happens if the incidence angle becomes  $0^{\circ}$ ?

#### **Solution:**

1) The wave is TE. For the total electric field in medium 1, the reflection coefficient is needed. The refraction angle is:

$$\begin{split} &\sqrt{\mu_{r1}\varepsilon_{r1}}\sin(\theta) = \sqrt{\mu_{r2}\varepsilon_{r2}}\sin(\theta_{t}) \Rightarrow \theta_{t} = 8.24^{\circ} \\ &\eta_{TE}^{1} = \eta_{0}\sqrt{\frac{\mu_{r1}}{\varepsilon_{r1}}}\frac{1}{\cos(\theta)} = 230.1\,\Omega \\ &\eta_{TE}^{2} = \eta_{0}\sqrt{\frac{\mu_{r2}}{\varepsilon_{r2}}}\frac{1}{\cos(\theta_{t})} = 190.5\,\Omega \\ &\Gamma = \frac{\eta_{TE}^{2} - \eta_{TE}^{1}}{\eta_{TE}^{2} + \eta_{TE}^{1}} = -0.094 \\ &\vec{E}_{t}(x,y,x) = \vec{E}_{i}(x,y,x) + \vec{E}_{r}(x,y,x) = j\mu_{x}e^{-j\frac{2\pi}{\lambda_{1}}[\cos(\theta)z + \sin(\theta)y]} + \Gamma j\mu_{x}e^{-j\frac{2\pi}{\lambda_{1}}[-\cos(\theta)z + \sin(\theta)y]} \\ &V/m \end{split}$$

where

$$\lambda_1 = \frac{c}{f\sqrt{\mu_{r1}\varepsilon_{r1}}} \approx 0.15 \text{ m}$$

2) The power density propagating in the second medium in the z direction is:

$$S_z^2 = S_z^1 (1 - |\Gamma|^2) = \frac{1}{2} \frac{|\vec{E}|^2}{\eta^1} \cos(\theta) (1 - |\Gamma|^2) = 2.2 \text{ mW/m}^2$$
  
where

$$\eta^1 = \eta_0 \sqrt{\frac{\mu_{r1}}{\varepsilon_{r1}}} = 188.5 \ \Omega$$

3) For orthogonal incidence:

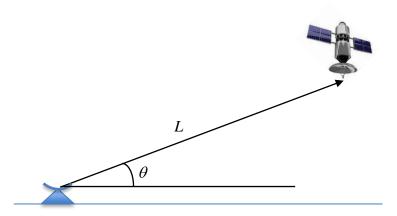
$$\eta^2 = \eta_0 \sqrt{\frac{\mu_{r2}}{\varepsilon_{r2}}} = 188.5 \ \Omega = \eta^1$$

Therefore, there is no reflected field in medium 1 and the whole power density crosses the interface.

#### **Problem 4**

Consider the uplink to a LEO satellite (pointing at the Earth center) from a ground station, operating at f = 30 GHz, whose antenna is pointed zenithally. Determine the uplink yearly availability to guarantee a minimum signal-to-noise ratio (SNR) of 6 dB at the satellite. The CCDF of the zenithal tropospheric attenuation is given by:

$$P(A_T^{dB}) = 100e^{-1.15A_T^{dB}}$$
 (A<sub>T</sub> in dB and P in %)



Additional assumptions and data:

- elevation angle  $\theta = 30^{\circ}$
- power transmitted from the ground  $P_T = 378 \text{ W}$
- radiation patter of the antennas (circular symmetry):  $f = [\cos(\phi)]^2$
- equivalent noise temperature emitted by the ground  $T_B = 200 \text{ K}$
- mean radiating temperature  $T_{mr} = 285 \text{ K}$
- gain of the antennas (on board the satellite and on the ground)  $G_T = G_R = 30 \text{ dB}$
- distance to the satellite L = 2400 km
- bandwidth of the receiver B = 2 MHz
- internal noise temperature of the receiver  $T_R = 310 \text{ K}$
- no additional losses in the transmitter and the receiver

#### **Solution**

The signal-to-noise ratio (SNR) is given by

$$SNR = \frac{P_T G_T f_T (\lambda/4\pi L)^2 G_R f_R A_S}{k [T_R + T_A] B}$$

where  $A_S$  is the slant path attenuation in linear scale,  $f_T = f_R = [\cos(90^\circ - \theta)]^2 = 0.25$ , k is the Boltzmann's constant  $(1.38 \times 10^{-23} \text{ J/K})$  and  $T_A$  is the equivalent antenna noise temperature. For this scenario, as the satellite points at the ground,  $T_A$  is therefore calculated as:

$$T_A = A_S T_B + T_{mr} (1 - A_S)$$

Therefore, imposing  $SNR_{min} = 6 \text{ dB} = 3.981$ :

$$SNR = \frac{P_T G_T f_T (\lambda / 4\pi L)^2 G_R f_R A_S}{k [T_R + A_S T_B + T_{mr} (1 - A_S)]B} = 3.981$$

Inverting the equation above to solve for  $A_S$ :

$$A_{S} = \frac{SNR_{min}(BkT_{R} + T_{mr}Bk)}{P_{T}G_{T}f_{T}(\lambda/4\pi L)^{2}G_{R}f_{R} - SNR_{min}(BkT_{B} - Bk)} = 0.0254$$

In dB:

$$A_S^{dB} = -10\log_{10}(A_S) = 15.948 \text{ dB}$$

The zenithal attenuation is:

$$A_T^{dB} = A_S^{dB} \sin(\theta) = 7.974 \text{ dB}$$

Using the CCDF expression, such an attenuation corresponds to an outage probability of roughly 0.01% in a year, i.e. to a yearly availability of 99.99%.