Satellite Communication and Positioning Systems - Prof. L. Luini, July $21{ }^{\text {st }}, 2022$


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## Problem 1

The figure below shows a bistatic radar system (carrier frequency $f_{1}=1 \mathrm{GHz}$ ), consisting of two LEO satellites (same orbit, satellite height above the ground $H=500 \mathrm{~km}$, elevation angle $\theta=45^{\circ}$ ). The radar extracts information on the atmosphere by measuring the difference between the power received along the direct path and the downlink to the ground station. Making reference to the right, where the electron content $\left(N_{e}\right)$ profile $\left(h_{\min }=100 \mathrm{~km}, h_{\max }=400 \mathrm{~km}, N_{m}=5 \times 10^{12} \mathrm{e} / \mathrm{m}^{3}, N\right.$ homogeneous horizontally) and the tropospheric specific attenuation profile ( $\alpha=0.01 e^{-h / 50}$, where $h$ is in km and $\alpha$, horizontally homogeneous, is in $\mathrm{dB} / \mathrm{km}$ ) are shown, and to the left side, where the simplified geometry is reported:

1) Calculate the difference of the power received along the two paths.
2) Will the difference between the power received along the two paths decrease or increase (compared to point 1) if the frequency changes to $f_{2}=25 \mathrm{MHz}$ ?
Assume: transmit power $P_{T}=100 \mathrm{~W}$; gain of all antennas (circular parabolic reflectors) $G=45 \mathrm{~dB}$; antenna radiation pattern $f=[\cos (\phi)]^{2}$, where $\phi$ is the angle between the specific direction and the antenna axis; perfect pointing for the downlink, on both ends.


## Solution

1) The power received along the direct path (free space) is simply given by:
$P_{R 1}=P_{T} G_{T} f_{T}(\lambda / 4 \pi D)^{2} G_{R} f_{R}$
where:
$D=\frac{2 H}{\operatorname{tg}(\theta)}$
$G_{T}=G_{R}=G$
$f_{T}=f_{R}=\cos (\theta)$
Therefore $\rightarrow P_{R 1}=14.2 \mu \mathrm{~W}$
Regarding the second path, the power density reaching the ground is:
$S=\frac{P_{T}}{4 \pi L^{2}} G A_{l}$
where $L=H / \sin (\theta)$ and $A_{l}$ is the tropospheric attenuation (the ionospheric one can be neglected given the carrier frequency in the GHz range). The zenithal tropospheric attenuation can be calculated as:
$A_{Z}=\int_{0}^{H} \alpha d h=\alpha_{0} \int_{H}^{H_{S}} e^{-\frac{h}{h_{0}}} d h=-\alpha_{0} h_{0}\left[e^{-\frac{h}{h_{0}}}\right]_{0}^{H}$
As $H \gg h_{0}=50 \mathrm{~km}$ :
$A_{Z}=-\alpha_{0} h_{0}\left[e^{-\frac{h}{h_{0}}}\right]_{0}^{\infty}=\alpha_{0} h_{0}=0.5 \mathrm{~dB}$
The slant path attenuation in linear scale is:
$A_{l}=10^{-\left(A_{Z} / \sin (\theta)\right) / 10}=\alpha_{0} h_{0}=0.85 \mathrm{~dB}$

The power received on the ground is:
$P_{R 2}=S A_{R X}=\frac{P_{T}}{4 \pi L^{2}} G A_{l} \frac{\lambda^{2}}{4 \pi} G=96.9 \mu \mathrm{~W}$
The differential power is:
$\Delta P_{R}=P_{R 2}-P_{R 1}=82.6 \mu \mathrm{~W}$
2) Using the second frequency (range of tens of MHz ), it is worth checking if the wave can actually cross the ionosphere. For the wave to avoid total reflection due to the ionosphere, the angle $\theta$ needs to be higher than $\theta_{\text {min }}$, determined as by:
$\cos \left(\theta_{\text {min }}\right)=\sqrt{1-\left(\frac{9 \sqrt{N_{\mathrm{m}}}}{f_{2}}\right)^{2}} \Rightarrow \theta_{\min }=53.6^{\circ}$
As $\theta<\theta_{\text {min }}$, the wave will be totally reflected and it will not be received by RX.

## Problem 2

A plane EM wave $\left(\left|\vec{E}_{T X}\right|=10 \mathrm{~V} / \mathrm{m}\right)$ at 30 GHz propagates across a layer of melting hydrometeors (layer thickness $h=1 \mathrm{~km}, \theta=45^{\circ}$ tilt angle), which is characterized by the following propagation constants (see sketch below):
$\gamma_{I}=0.6413+j 6286761 / \mathrm{km}$
$\gamma_{I I}=0.6413+j 628674.431 / \mathrm{km}$
Both the transmitter (TX) and the receiver (RX) employ linear antennas; the TX antenna is horizontal linear; as for the RX antenna, see point 2). For this link:

1) Determine the wave polarization in front of the receiver $R X$ (no need to specify the tilt angle for a linear polarization, nor the rotation direction for a circular/elliptical polarization)
2) Based on point 1 , determine the best antenna to maximize the received power.


## Solution

1) As is clear from the propagation constants, the hydrometeor slab induces the same attenuation on both I and II components of the wave; on the other hand, the differential phase shift is:
$\Delta \beta=\beta_{I I}-\beta_{I}=-1.57=-\pi / 2 \mathrm{rad} / \mathrm{km}$
The TX antenna emits a vertical polarization, so the amplitude of components I and II are equal and given by $\left|\vec{E}_{T X}\right| \cos \left(45^{\circ}\right)=7.07 \mathrm{~V} / \mathrm{m}$. Therefore, in front of RX, the two components I and II will have: 1) same amplitude (attenuated by the slab); 2) a differential phase shift of $\pi / 2 \mathrm{rad} / \mathrm{km} \rightarrow$ the polarization will be circular.
2) Given the circular polarization determined at point 1 , a circular polarized antenna is the best choice.

## Problem 3

The figure below shows a dual-frequency GNSS RX (positioned at average mean sea level altitude) tracking two GPS satellites at different elevation angles $\left(\theta_{1}=45^{\circ}, \theta_{2}=30^{\circ}\right.$, satellites height above the ground $H=20180 \mathrm{~km}$ ). Making reference to the right, where the refractivity profile ( $N=N_{0} e^{-h / 40000} \mathrm{ppm}$, where $h$ is in m and $N$ is homogeneous horizontally) are shown, and to the left side, where the simplified geometry is reported (flat Earth): calculate the value of $N_{0}$ and the receiver clock bias, knowing that the pseudorange for S1 and S2 measured by the GNSS RX are $\rho_{1}=28538868 \mathrm{~m}$ and $\rho_{2}=40360058 \mathrm{~m}$, respectively. To this aim, assume perfect code synchronization.


## Solution

The two slant paths can be easily calculated as:
$L_{1}=\frac{H}{\sin \left(\theta_{1}\right)} \approx 28538829 \mathrm{~m}$
$L_{2}=\frac{H}{\sin \left(\theta_{2}\right)} \approx 40360000 \mathrm{~m}$
The pseudorange is given by:
$\rho=L+\frac{d^{I}}{\sin (\theta)}+\frac{d^{T}}{\sin (\theta)}+d^{C / A}+c t_{B}$
Assuming perfect code synchronization $\rightarrow d^{C / A}=0 \mathrm{~m}$; also, for a dual-frequency receiver $\rightarrow d^{I}=$ 0 m .
$\rho=L+\frac{d^{T}}{\sin (\theta)}+c t_{B}$
Given the exponential trend of the refractivity profile, the zenital tropospheric delay can be calculated as:
$d^{T}=-h_{0} N_{0} 10^{-6}\left[e^{-\frac{h}{h_{0}}}\right]_{0}^{H}$
Considering the satellite height $H \gg h_{0}=40 \mathrm{~km}$ :
$d^{T}=h_{0} N_{0} 10^{-6}$
We have two equations with two unknowns ( $d^{T}$ and $t_{B}$ ). Taking the difference of $\rho_{2}$ and $\rho_{1}$ :
$\rho_{2}-\rho_{1}=\left(L_{2}-L_{1}\right)+d^{T}\left[\frac{1}{\sin \left(\theta_{2}\right)}-\frac{1}{\sin \left(\theta_{1}\right)}\right]$
as the clock bias is the same for both pseudoranges. As a result:
$d^{T}=\frac{\left(\rho_{2}-\rho_{1}\right)-\left(L_{2}-L_{1}\right)}{\left[\frac{1}{\sin \left(\theta_{2}\right)}-\frac{1}{\sin \left(\theta_{1}\right)}\right]}=32 \mathrm{~m}$
Therefore:
$N_{0}=\frac{d^{T}}{h_{0} 10^{-6}}=800 \mathrm{ppm}$
Finally:
$\rho_{1}=L_{1}+\frac{d^{T}}{\sin \left(\theta_{1}\right)}+c t_{B}$
$t_{B}=\frac{1}{c}\left[\rho_{1}-L_{1}-\frac{d^{T}}{\sin \left(\theta_{1}\right)}\right]=-20 \mathrm{~ns}$

## Problem 4

Consider a satellite link, implementing adaptive coding and modulation, which adapts the data rate depending on the link signal to noise ratio (SNR). The link elevation angle is $\theta=45^{\circ}$ and the link operates at $f=20 \mathrm{GHz}$. The Complementary Cumulative Distribution Function (CCDF) of the zenithal tropospheric attenuation is given by:

$$
P\left(A_{d B}^{Z}\right)=100 e^{-0.545 A_{d B}^{Z}} \quad\left(A_{d B}^{Z} \text { in } \mathrm{dB} \text { and } P \text { in } \%\right)
$$

Determine the yearly time for which $10 \mathrm{Mbit} / \mathrm{s}$ can be guaranteed to the user.


| $15 \mathrm{~dB}<\mathrm{SNR} \leq 20 \mathrm{~dB}$ | $D=20 \mathrm{Mbit} / \mathrm{s}$ |
| :---: | :---: |
| $10 \mathrm{~dB}<\mathrm{SNR} \leq 15 \mathrm{~dB}$ | $D=10 \mathrm{Mbit} / \mathrm{s}$ |
| $\mathrm{SNR} \leq 10 \mathrm{~dB}$ | $D=1 \mathrm{Mbit} / \mathrm{s}$ |

Additional assumptions and data:

- both antennas pointed optimally
- disregard the cosmic background radiation
- power transmitted by each satellite $P_{T}=100 \mathrm{~W}$
- mean radiating temperature $T_{m r}=290 \mathrm{~K}$
- gain of both antennas $G=35 \mathrm{~dB}$
- satellite altitude $H=600 \mathrm{~km}$
- bandwidth of the receiver: $B=100 \mathrm{MHz}$
- internal noise temperature of the receiver: $T_{R}=300 \mathrm{~K}$


## Solution

The signal-to-noise ratio (SNR) is given by
$S N R=\frac{P_{T} G_{T} f_{T}(\lambda / 4 \pi L)^{2} G_{R} f_{R} A}{k\left[T_{R}+T_{m r}(1-A)\right] B}$
where $k$ is the Boltzmann's constant $\left(1.38 \times 10^{-23} \mathrm{~J} / \mathrm{K}\right), f_{T}=f_{R}=1, L=H / \sin (\theta), G_{R}=G_{T}=G$ (same antenna features). The target data rate is obtained for $10 \mathrm{~dB}<\mathrm{SNR} \leq 15 \mathrm{~dB}$ : to calculate the yearly time for which $10 \mathrm{Mbit} / \mathrm{s}$ can be guaranteed to the user, the threshold $\mathrm{SNR}_{\text {min }}=10 \mathrm{~dB}$ is used. Therefore, the above equation can be solved for $A$ (slant path attenuation in linear scale):
$A=\frac{S N R_{\min } k B\left(T_{R}+T_{m r}\right)}{P_{T} G^{2}(\lambda / 4 \pi L)^{2}+S N R_{\min } k B T_{m r}}=0.0041$
The slant path attenuation in dB is:
$A_{d B}=-10 \log _{10}(A)=23.9 \mathrm{~dB}$
The zenithal path attenuation in dB is:
$A_{d B}^{Z}=A_{d B} \sin (\theta)=16.9 \mathrm{~dB}$
Using such a value in the CCDF of the tropospheric attenuation:
$P=0.01 \%$, which corresponds to approximately 0.89 hours in a year. Therefore, $10 \mathrm{Mbit} / \mathrm{s}$ can be guaranteed for $99.99 \%$ of the yearly time, i.e. always but 0.89 hours ( 53 minutes) in a year.

